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# Direct numerical simulation on the effects of surface slope and skewness on rough-wall turbulence

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This paper presents direct numerical simulation (DNS) results of turbulent flows over systematically varied rough surfaces. Three-dimensional irregular rough surfaces with varying effective slope and skewness factor and fixed roughness height scales were considered in the study. The skewness factor characterizes whether the surface of interest has a peak-dominated or valley-dominated nature, whereas the effective slope measures the wavelength of the surface undulations or solidity of the roughness elements. The influence of these two topological parameters on the friction drag at rough surfaces was investigated. Downward shifts in the inner-scaled mean velocity, which quantify an increase in the friction drag, were found to be larger for surfaces with a positive skewness factor, and this trend was found to be more pronounced as the effective slope increased. In addition, the downward shift value steeply increased with increases in the effective slope, while the dependence weakened when the effective slope was larger than a certain threshold value. The physical mechanism behind the increase in the roughness function was investigated by analyzing the momentum budgets. It was revealed that the viscous drag dominantly contributes to the roughness function when the effective slope value is small, whereas the contribution by the pressure drag progressively increases with the effective slope. We also found that for surfaces with larger effective slope consisting of relatively shorter wavelength undulations, the Reynolds shear stress tends to be reduced because the wall roughness prevents the formation of quasi-streamwise elongated vortices suppressing the turbulent near-wall cycles. This acts as a negative contribution to the roughness function, and the two competing effects (of the increase in pressure drag and decrease in Reynolds shear stress) weaken the dependence of the effective slope value on the roughness function. Further analysis was conducted to better understand how the surface slope and skewness factor values affect the mean flow field, modifying the pressure and viscous drag forces.

## I. INTRODUCTION

Rough-wall turbulence has received significant attention over the past decades because surfaces in engineering systems are seldom hydraulically smooth but typically have roughness. The surface of a turbine blade becomes rough over time owing to deposition, pitting, and spallation that occur while operating in harsh conditions. Similarly, time-related deterioration, including erosion, corrosion, and organic and inorganic fouling processes, generate roughness in ship hulls, oil pipelines, and heat exchangers. In most situations, the flow in these systems is turbulent, and the wall roughness is not within the viscous sublayer but protrudes into the logarithmic layer, leading to a significant increase in friction drag. Consequently, this results in substantial performance degradation of the aforementioned systems. Hence, predicting turbulence modification due to wall roughness becomes an important prerequisite for optimal engineering design and machine maintenance. To quantify such surface roughness effects, the equivalent roughness  $k_s$  and the roughness function  $\Delta U^+$  have been widely employed. The equivalent roughness is defined as the size of sand grain that yields the same skin friction coefficient as the surface of interest, and  $\Delta U^+$  is the downward shift value in the inner-scaled mean velocity profile in the logarithmic region.

One of the important characteristic that has a significant impact on surface roughness effects is the solidity of the roughness elements. Earlier works on the influence of solidity for two-dimensional surface roughness were conducted by

Bettermann<sup>1</sup>, Dvorak<sup>2</sup>, and Dirling<sup>3</sup>. Dirling<sup>3</sup> provided correlations for  $k_s$  based on two parameters: a solidity parameter and a shape parameter. The solidity parameter was represented as the ratio of the average element spacing to roughness height, while the shape parameter accounted for the frontal area and the windward wetted surface area of a single roughness element. An alternative expression including the solidity parameter  $\Lambda$  was proposed by Sigal and Danberg<sup>4</sup>, which was based on the reference surface area before the addition of roughness and the total frontal area over the rough surface. Another representation for solidity, which is significantly easier to define for three-dimensional irregular roughness, is the effective slope ( $ES$ ) proposed by Napoli *et al.*<sup>5</sup>; it is defined as follows:

$$ES = \frac{1}{L_x L_z} \int_x \int_z \left| \frac{\partial h(x, z)}{\partial x} \right| dx dz, \quad (1)$$

where  $h(x, z)$  is the roughness height, and  $L_x$  and  $L_z$  are the respective streamwise and spanwise lengths of the rough surface.  $ES$  is defined as the average value of the magnitude of the slope of the roughness corrugation, and it can be shown that  $ES$  is double the value of the solidity parameter:  $ES = 2\Lambda^{5-7}$ . To demonstrate the ability of  $ES$  in predicting surface roughness effects, Napoli *et al.*<sup>5</sup> performed a large eddy simulation (LES) of two-dimensional irregular corrugated walls and found that the roughness function  $\Delta U^+$  increased with the  $ES$  value when the  $ES$  value was lower than a certain threshold. They also showed that flow separation behind the roughness crest occurred more frequently as the  $ES$  value increased, and this increased the contribution of the pressure drag to the total friction drag. A simi-

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lar trend of dependence of the  $ES$  value on  $\Delta U^+$  was confirmed via experiments on closely packed pyramids<sup>8</sup> and direct numerical simulation (DNS) studies on sinusoidal wavy walls<sup>9</sup>, randomly distributed semi-ellipsoid/cone roughness<sup>10</sup>, and a variety of real rough surfaces such as concrete, grit-blasted, and graphite<sup>6</sup>. Using large amounts of LES data, De Marchis<sup>11</sup> proposed a new mathematical logarithmic law that could reasonably predict  $\Delta U^+$  based on the  $ES$  value. Yuan and Jouybari<sup>12</sup> provided a physical interpretation of a relatively smaller drag for a surface with a small  $ES$  value. They stated that the occurrence of roughness wakes was decreased for the surface with the small  $ES$  value, thus reducing the pressure drag. In addition, they reported that the energy redistribution to the wall-normal Reynolds stress was attenuated for the surface with small  $ES$  value, owing to the reduction in the form-induced shear and mixing-layer turbulence activity. Extensive investigations on the effects of solidity were also conducted by MacDonald *et al.*<sup>7</sup> as well as Leonardi and Castro<sup>13</sup>, who also focused on surfaces with densely distributed roughness elements. MacDonald *et al.*<sup>7</sup> showed that although  $\Delta U^+$  first increased with the  $ES$  value as reported in Napoli *et al.*<sup>5</sup>, it decreased in the dense roughness regime with larger  $ES$  values. They stated that this trend was consistent with the observations in the so-called  $d$ -type roughness<sup>14</sup>, and the driving mechanism was attributed to a reduction in the Reynolds shear stress that was predominantly due to the near-wall turbulence cycle being pushed away from the rough wall.

The other important characteristic parameter for determining surface roughness effects is the skewness factor ( $Sk$ ). The skewness factor, which is defined by the statistical moments of the surface elevation, quantifies the asymmetry of the probability density function (PDF) of the roughness height elevation. The skewness factor  $Sk$  is defined as follows:

$$Sk = \frac{1}{h_{rms}^3 L_x L_z} \int_x \int_z (h(x, z) - h_m)^3 dx dz, \quad (2)$$

where  $h_m$  is the mean roughness height, and  $h_{rms}$  is the root-mean-square roughness height:

$$h_{rms}^2 = \frac{1}{L_x L_z} \int_x \int_z (h(x, z) - h_m)^2 dx dz. \quad (3)$$

The skewness factor characterizes whether the surface of interest has a peak-dominated or valley-dominated nature: a surface with a positive  $Sk$  value is peak-dominated whereas a surface with a negative value is valley-dominated. An earlier attempt to relate the statistical moments of the surface elevation to the equivalent roughness was made by Musker<sup>15</sup>, and many correlations toward predicting the surface roughness effects using the skewness factor have been proposed<sup>6,10,16–18</sup>. A large amount of experimental data for several types of rough surfaces, including packed spheres, sandpaper, gravel, honed pipes, and scratched plates, suggested that the surface roughness effects were enhanced with an increasing  $Sk$  value when  $Sk > -1$ <sup>16</sup>. This has also been corroborated by recent DNS studies<sup>6,10,18–20</sup>. Measurements of a systematically varied surface by Flack *et al.*<sup>21</sup> revealed that a dramatic increase in friction drag occurred when the  $Sk$  value changed from a negative value to zero, while a change from zero to a positive

value caused a modest increase in friction drag. A similar experimental study by Flack *et al.*<sup>22</sup> suggested the need for separate predictive equations for surfaces with positive and negative skewness values. However, Flack *et al.*<sup>22</sup> as well as Busse *et al.*<sup>20</sup> reported that the transitional behavior toward the fully rough regime was almost independent of the  $Sk$  value. Jelly and Busse<sup>23</sup> discussed surface roughness effects using DNS for three roughness topographies: a Gaussian surface, a peaks-only surface, and a pits-only surface characterized by zero, positive, and negative  $Sk$  values, respectively. They reported that the peaks-only surface with a positive  $Sk$  value yielded a  $\Delta U^+$  comparable to that of the Gaussian surface, while the  $\Delta U^+$  for the pits-only surface with a negative  $Sk$  value was much smaller. The significantly larger  $\Delta U^+$  value for the peaks-only surface in comparison with that for the pits-only surface was mainly attributed to the velocity offset at the roughness crest, which represented the integral effects of pressure and the viscous effects below the roughness crest.

As mentioned above, systematic investigations on  $Sk$  or  $ES$  have been carried out experimentally<sup>8,22,24</sup> and numerically<sup>6,7,10,18,23,25</sup>, and the importance of  $Sk$  and  $ES$  values for determining surface roughness effects has been well established. Meanwhile, it has also been reported that many other important characteristic parameters need to be taken into account when predicting surface roughness effects (e.g., the Kurtosis defined as the fourth-order statistical moment<sup>10,15</sup>, diversity of roughness peak heights<sup>10,26</sup>, and streamwise correlation length<sup>6</sup>, as well as roughness height scales such as mean peak height, peak-to-valley height, and roughness height amplitude). This causes difficulties in isolating the influence of each characteristic parameter on surface roughness effects, and remains a major obstacle in appropriately accounting for these characteristic parameters predicting the effects of surface roughness.

To resolve these difficulties, we investigated the individual effects of skewness  $Sk$  and slope (solidity)  $ES$  on surface roughness effects through DNSs of turbulence over three-dimensional irregular rough surfaces, in which the  $Sk$  and  $ES$  values were systematically varied while the other characteristic parameters remained fixed. Further, to better understand how these parameters ( $Sk$  and  $ES$ ) affect the flow structure and momentum transport mechanisms, this study also analyzed the double-averaged (spatial and Reynolds) momentum equation.

## II. FLOW CONDITIONS

As in the previous DNS studies<sup>10,13,18,27,28</sup>, we chose a rough-walled open channel flow configuration as shown in Fig. 1 in which a flow is periodic in the streamwise and spanwise directions whereas the slip boundary conditions are applied at the top wall. Although the flow near the slip wall is not identical to that near the symmetry plane in full channel flow, the choice of the so-called open channel flow allows us to simulate turbulence modification near the rough wall with fewer computational resources. The computational do-

main size ( $L_x, L_y, L_z$ ) is  $(6\delta, \delta, 3\delta)$  in the streamwise, wall-normal, and spanwise directions, respectively, where  $\delta$  is the half-channel height. The flow is driven by a constant streamwise pressure difference, and the friction Reynolds number based on the effective half-channel height,  $\delta_e = \delta - h_m$  with  $h_m$  being the mean rough surface height, is fixed at  $\text{Re}_\tau = 600$ . Here, the friction velocity  $u_\tau$  is given by the averaged wall-shear stress  $u_\tau = \sqrt{\tau_w/\rho}$ , which is computed by the streamwise momentum balance between the pressure drop  $\Delta P$  and the wall-shear stress:

$$\Delta P S_{yz} = \tau_w L_x L_z, \quad (4)$$

where  $S_{yz} = V_f/L_x$  is the averaged fluid phase  $y-z$  plane area with  $V_f$  being the volume of the fluid phase and can be simply written as  $S_{yz} = \delta_e L_z$  (Kuwata and Kawaguchi<sup>18</sup>). Hence, we can obtain the wall-shear stress as follows:

$$\tau_w = \delta_e \frac{\Delta P}{L_x}. \quad (5)$$

It should be noted that the  $\tau_w$  given by Eq.(5) is equivalent to the  $x-z$  plane-averaged total shear stress extrapolated at the location of the mean roughness height. This definition is consistent with those employed in previous DNS studies<sup>9,10,18,29</sup>. In this study, we considered the lattice Boltzmann equation for the governing equation, which is proven to recover the continuity and Navier–Stokes equations in second-order accuracy in space and time. Although there are several possible choices for the LBM models, we used the three-dimensional twenty-seven (D3Q27) discrete velocity multiple-relaxation-time lattice Boltzmann method (LBM)<sup>30</sup>. This DNS approach has been validated against turbulent channel flows, and no perceptible differences were found between the mean velocity, second moments, energy spectra, and turbulent budget terms of the solutions from the LBM and spectral methods<sup>30</sup>. Further, the present LBM-DNS approach was successfully applied in turbulent flows over complicated rough surfaces<sup>18,28,31,32</sup>. The LBM employs a regular grid with equal spacing in which non-body-fitted Cartesian mesh to describe the rough-wall geometry. This feature reduces numerical errors arising from a coordinate transformation procedure; however, it requires prohibitively high computing resources for handling the entire computational domain with sufficiently fine grid resolution. Hence, the local grid refinement technique proposed by Kuwata and Suga<sup>33</sup> was adopted to allocate the doubly refined grid near the rough-wall region for  $0 < y < 0.34\delta$ . The grid resolution near the rough-wall region was determined such that the grid spacing in wall units is less than 2.0, as in the previous lattice Boltzmann DNS studies<sup>18,28,29,32</sup> where the wall unit is defined as  $u_\tau/\nu$  with  $\nu$  denoting the kinematic viscosity. As a result, the total number of grid points was approximately 293 million. To handle the rough wall geometry, we employed the linear interpolated bounce-back method, as in the previous DNS studies<sup>18,28,29,32</sup>. This method is based on the bounce-back rule and accurately imposes no-slip boundary conditions to the rough surface with second-order accuracy<sup>34</sup>. The numerical setup, including the domain size and grid resolution, is comparable to those employed in previous DNS studies on rough-walled open chan-

TABLE I. Characteristic parameters of rough surfaces;  $Sk$  is the skewness factor,  $ES$  is the effective slope,  $h_{rms}$  is the root-mean-square roughness,  $h_a$  is the surface height amplitude, and  $h_t$  is the mean peak-to-valley height.

$Sk$	$ES$	$h_{rms}^+$	$h_a^+$	$h_t^+$	$h_{rms}/\delta_e$
$\pm 0.53$	0.1, 0.2, 0.4, 0.6	8.4	6.7	49	0.014

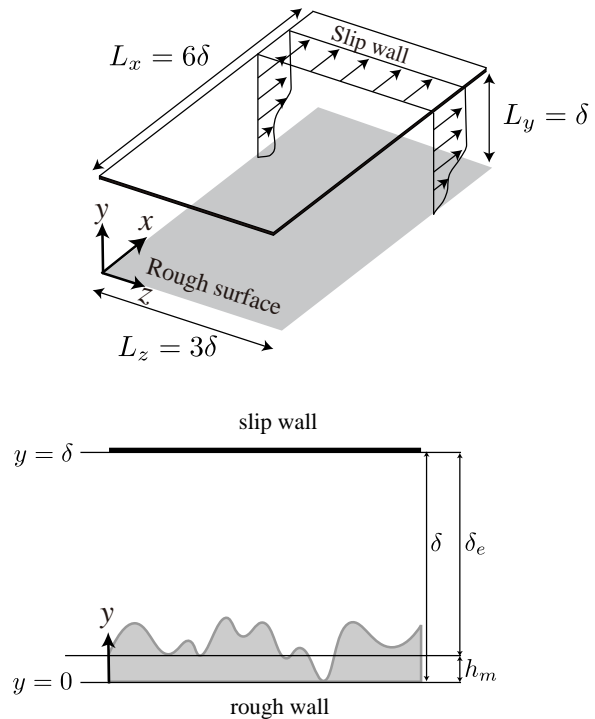


FIG. 1. Schematic of flow geometry of a rough-walled open channel flow.

nel flows<sup>18,28,31,35</sup>; therefore, it has been confirmed to be sufficient for capturing the full spectrum of scales, ranging from dissipative fine eddies to large-scale fluctuations.

### III. ROUGH SURFACES

In this study, we considered 8 rough surfaces in which the  $Sk$  and  $ES$  values were systematically varied while the roughness height scales remained fixed. As mentioned previously, the skewness factor  $Sk$ , defined in Eq. (2), measures whether the surface of interest has a valley-dominated or a peak-dominated nature, whereas the effective slope  $ES$ , defined in Eq. (1), measures the wavelength of the surface undulations or solidity of the roughness elements. In what follows, we describe the procedure used for generating the rough surfaces. The surface height  $h(x, z)$  was first generated by superimposing differently sized hyperbolic shape roughness elements. The height of a single hyperbolic shape roughness element  $h_N(x, z)$  is defined as a rotating body of the hyperbolic

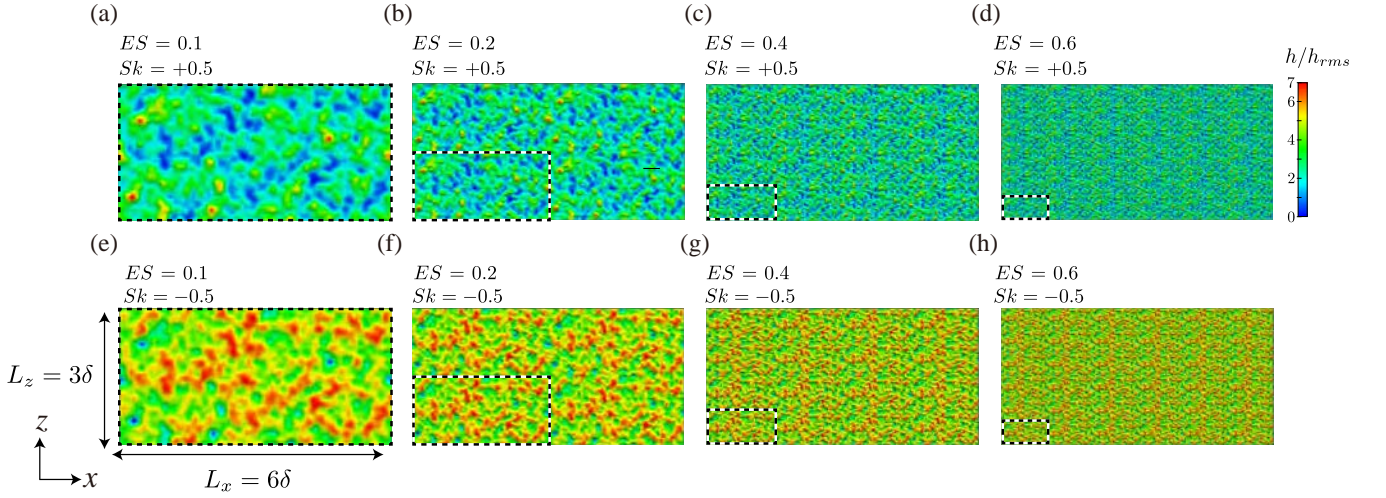


FIG. 2. Geometry of the rough surfaces together with the effective slope value in Eq. (1) and the skewness factor in Eq. (2). The surface enclosed by a dashed line indicates a single unit tile.  $2 \times 2$ ,  $4 \times 4$ ,  $6 \times 6$  tiles are used for (b,f), (c,g), and (d,h), respectively.

function:

$$h_N(x, z) = (A_N \delta) \operatorname{sech} \left( B_N \frac{\sqrt{(x - x_N)^2 + (z - z_N)^2}}{\delta} \right), \quad (6)$$

where  $x_N$  and  $z_N$  represent the center position of the roughness element, and the shape parameters  $A_N$  and  $B_N$  determine the height and width of a roughness element, respectively. Then, the rough surface height is given by superimposing the roughness elements as follows:

$$h(x, z) = \sum_N h_N(x, z), \quad (7)$$

where  $N$  denotes the number of roughness elements, and  $x_N$ ,  $z_N$ ,  $A_N$ , and  $B_N$  are all given by random numbers. The surface characteristics can be controlled through the number of roughness elements  $N$  and by imposing the upper and lower limits for the  $A_N$  and  $B_N$  values. The values for  $N$ ,  $A_N$ , and  $B_N$  were adjusted by trial and error such that the  $Sk$  and  $ES$  values were close to the target values of  $Sk = +0.53$  and  $ES = 0.1$ . The resulting values for these parameters are  $N = 10000$ ,  $1.2 \times 10^{-3} \leq A_N \leq 1.2 \times 10^{-2}$ , and  $14.4 \leq B_N \leq 28.8$ , and the generated rough surface is shown in Fig. 2(a) including the  $Sk$  and  $ES$  values. Next, we describe how the  $ES$  and  $Sk$  values were systematically changed. The sign of the  $Sk$  value was changed by reversing the surface height as  $h_p - h(x, z)$  with  $h_p$  being the maximum roughness peak. This transformed the surface peaks of the original rough surface into surface valleys in the reversed surface; this is clearly shown in Fig. 2(a,e), where the positions of the surface peaks in Fig. 2(a) correspond to the positions of the surface valleys in Fig. 2(e). Notably, this transformation did not affect the surface height scales and the  $ES$  value. The  $ES$  value was increased by reducing the surface width in the streamwise and spanwise directions while preserving the surface height. This procedure effectively altered the wavelength of the surface undulation but kept the roughness height scales fixed. For example, as

shown in Fig. 2 (a,b), we reduced the width of the original surface (Fig. 2 (a)) in the streamwise and spanwise directions by a factor of 2; and when  $2 \times 2$  reduced surfaces are used, the  $ES$  value is doubled, as shown in Fig. 2 (b). Following this procedure, the  $ES$  value was increased from 0.1 to 0.6; this range of  $ES$  values covers wavy surface to rough surface regimes<sup>8</sup>. The characteristic parameters, including the values of  $Sk$ ,  $ES$ , and  $h_{rms}$ ; the surface height amplitude  $h_a$ ; and the mean peak-to-valley height  $h_l$ , are summarized in Table I, where  $h_l$  is computed by partitioning the surface in Fig. 2(a) into  $5 \times 5$  tiles of equal size<sup>6</sup>. It should be noted that unlike other systematic studies on the characteristic parameters of a rough surface, this procedure only modified the  $ES$  and  $SK$  values and strictly preserved the roughness height scales (e.g., root-mean-square roughness, mean peak-to-valley height, and surface height amplitude) and the kurtosis; this allows us to investigate the isolated effects of  $ES$  and  $Sk$  values on turbulence. However, on the negative side, this procedure reluctantly imposes the repetition of the surface height in  $x$  and  $z$  directions for cases with  $ES = 0.2, 0.4$ , and  $0.6$ . Thus, these surfaces cannot be regarded as perfectly irregular rough surfaces, and the effects of the roughness arrangement may appear; these aspects are discussed in §V. Note that, as shown in Fig.3, the grid resolution is sufficiently fine to accurately reproduce the surface geometry, even for the case with  $ES = 0.6$ , which has the steepest surface undulations among the presently tested cases.

#### IV. AVERAGING PROCEDURE

To statistically discuss turbulence near the rough wall, where a time-averaged variable changes in the  $x - z$  plane due to the inhomogeneous nature of the surface, we focus on a variable averaged over space and time in the following discussion. For spatial averaging, we introduce superficial  $x - z$

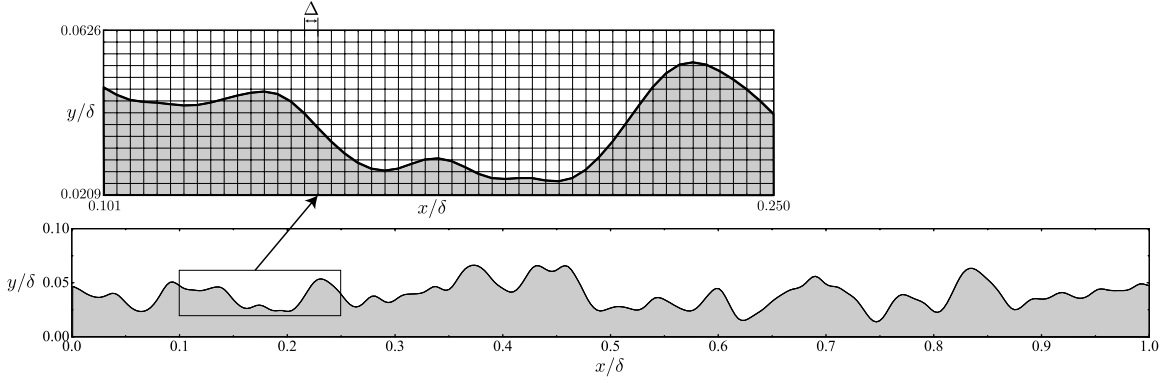


FIG. 3. Grid resolution near the rough surface for case with  $ES = 0.6$  and  $Sk = +0.53$ . Here,  $\Delta$  denotes the near-wall grid spacing.

plane-averaging for a variable  $\phi(x, y, z)$  as follows:

$$\langle \phi \rangle(y, t) = \frac{1}{S} \int_x \int_z \phi(x, y, z, t) dx dz, \quad (8)$$

where the plane area  $S = L_x L_z$ . We can also define the variable averaged over a fluid phase  $x - z$  plane as the intrinsic (fluid phase) averaged value:

$$\langle \phi \rangle^f(y, t) = \frac{1}{S_f} \int_x \int_z \phi(x, y, z, t) dx dz, \quad (9)$$

where  $S_f$  is the fluid phase  $x - z$  plane area, and a relation exists between the superficial and intrinsic plane-averaged values as:  $\langle \phi \rangle = \varphi \langle \phi \rangle^f$ , with the plane porosity  $\varphi = S_f/S$ . The variable  $\phi(x, y, z, t)$  can be decomposed into a contribution from an intrinsic averaged value  $\langle \phi \rangle^f(y, t)$  and a deviation from the intrinsic averaged value  $\tilde{\phi}(x, y, z, t)$  referred to as the dispersion, as follows:

$$\phi(x, y, z, t) = \langle \phi \rangle^f(y, t) + \tilde{\phi}(x, y, z, t). \quad (10)$$

As the flow variables also fluctuate in time, the Reynolds decomposition is introduced to decompose a variable into a Reynolds (ensemble) averaged value  $\bar{\phi}(x, y, z)$  and its fluctuation  $\phi'(x, y, z, t)$  as

$$\phi(x, y, z, t) = \bar{\phi}(x, y, z) + \phi'(x, y, z, t), \quad (11)$$

For Reynolds averaging, the statistical properties were assembled over a period of  $250T$ , where  $T = L_x/U_b$  is the flow-through time and  $U_b$  is the bulk mean velocity.

## V. MEAN VELOCITY AND ROUGHNESS FUNCTION

We first present profiles of the streamwise mean velocity to discuss how the  $ES$  and  $Sk$  values affect the mean flows. Profiles of the inner-scaled streamwise mean velocity  $\langle \bar{u} \rangle^+$  are presented in Fig. 4, where the effective wall-normal distance  $y_e$ :

$$y_e = \int_0^y \varphi dy, \quad (12)$$

proposed by Kuwata and Kawaguchi<sup>18</sup> is used as the distance from the rough wall. Notably,  $y_e$  becomes zero at the bottom of the deepest valley but returns to  $y - h_m$  above the maximum roughness crest  $y > h_p$ <sup>18</sup>. For comparison, the DNS result for the smooth-wall case<sup>36</sup> is also included. As the present DNS considers the open-channel flows, the present smooth wall result deviates from the reference data for the full channel flow as it approaches the channel center. Nevertheless, the present DNS is in perfect agreement with the reference data near the wall region, thus indicating that the present grid resolution is sufficient to resolve the viscous sublayer near the wall.

The figure confirms that the profiles of  $\langle \bar{u} \rangle^+$  for the rough-wall cases are shifted downward, which is due to an increase in the skin friction coefficient of the rough wall. The roughness function  $\Delta U^+$ , evaluated as the difference in  $\langle \bar{u} \rangle^+$  at  $y_e^+ \simeq 100$  with the smooth-wall case results as in Kuwata and Kawaguchi<sup>18</sup>, attains a maximum value of 7.3 ( $Sk = +0.53$  and  $ES = 0.6$ ), followed by 7.0 ( $Sk = +0.53$  and  $ES = 0.4$ ) and 5.3 ( $Sk = -0.53$  and  $ES = 0.6$ ). This indicates that the simulated flows are in the transitionally rough regime except the case with  $Sk = +0.53$  and  $ES = 0.6$ , judging from the criterion of the fully rough regime<sup>37</sup>  $\Delta U^+ \gtrsim 7(k_s^+ > 70)$ . For the case with  $Sk = +0.53$  and  $ES = 0.6$ , where the flow is expected to be in the onset of the fully rough regime, the equivalent roughness height  $k_s^+$  is estimated by the following relation<sup>16</sup>:

$$k_s^+ = \exp[\kappa(8.5 - B + \Delta U^+)]. \quad (13)$$

In the above equation,  $k_s^+ = 78$  when the von Kármán constant  $\kappa = 0.41$  and the log-law intercept for a smooth  $B = 5.2$  are used. Interestingly, this value reasonably agrees with the empirical correlation from the data for real rough surfaces<sup>16</sup>  $k_s = 4.43h_{rms}(1 + Sk)^{1.37}$  within 16%. This suggests that the generated surface yields hydraulic roughness effects comparable to real rough surfaces, despite the fact that the same surface geometry repeatedly appears in the streamwise and spanwise directions as mentioned in §III.

As for the influence of  $Sk$ , the figure confirms that  $\Delta U^+$  is larger for cases with the positive skewness value  $Sk = +0.53$ , which is consistent with the observations of previous studies<sup>6,10,16,18,23</sup>. Another observation from Fig. 4 is that  $\Delta U^+$  increases with the  $ES$  value:  $\Delta U^+$  substantially in-



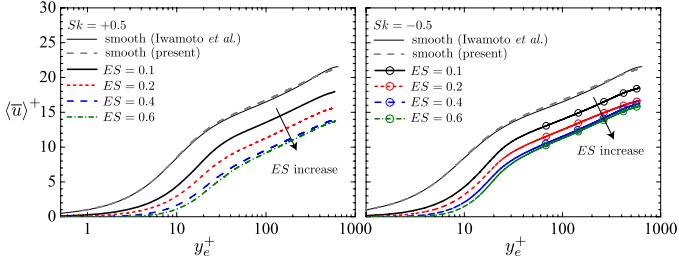


FIG. 4. Inner-scaled streamwise mean velocity profile. The DNS result for the smooth-wall case<sup>36</sup> is included.

creases as  $ES$  increases from 0.1 to 0.4, while this trend slows down when  $ES$  increases from 0.4 to 0.6.

The aforementioned is evident from Fig. 5, where  $\Delta U^+$  is plotted against the  $ES$  value. Also shown are the LES data for two-dimensional irregularly corrugated walls<sup>5</sup>, the DNS data for three-dimensional sinusoidal roughness<sup>9</sup>, and the experimental data for closely packed pyramids<sup>8</sup>. Note that the inner-scaled roughness height scales for the three-dimensional sinusoidal roughness<sup>9</sup> ( $h_{rms}^+ \simeq 5$ ) and the closely packed pyramids<sup>8</sup> ( $h_{rms}^+ \simeq 8.5$ ) are rather close to the values in the present study ( $h_{rms}^+ = 8.4$ ), whereas it is much larger in the two-dimensional irregularly corrugated wall<sup>5</sup> ( $h_a^+ = 19.5$ ) compared to the present study ( $h_a^+ = 6.7$ ). Figure 5 shows that  $\Delta U^+$  steeply increases with the  $ES$  value for the wavy surface regime<sup>8</sup> ( $ES < 0.3$ ); however, this trend progressively weakens as the  $ES$  value further increases, which is consistent with the findings of previous studies<sup>5,8</sup>. The values of  $\Delta U^+$  in the present study are close to the values reported by Schultz and Flack<sup>8</sup> and Chan *et al.*<sup>9</sup>, but deviate significantly from the results of Napoli *et al.*<sup>5</sup>, in which the inner-scaled roughness height scales are significantly larger. This supports the findings by Chan *et al.*<sup>9</sup> that  $\Delta U^+$  depends on some measure of the viscous roughness height as well as the  $ES$  value. Another important observation from the figure is that the dependence of the  $ES$  value is generally similar irrespective of the  $Sk$  value; however, the influence of  $Sk$  on  $\Delta U^+$  is more pronounced for a surface with a large  $ES$  value: the difference in  $\Delta U^+$  is 20% when  $ES = 0.1$  whereas the value is doubled when  $ES = 0.6$ .

## VI. MOMENTUM BUDGET

To better understand the physical mechanisms of the increase in friction drag, this section provides a discussion on momentum transfer with the help of the double-averaging theory. Applying the spatial- ( $x-z$  plane) and Reynolds-averaging operators to the Navier–Stokes equation for incompressible flows, we can obtain the double-averaged momen-

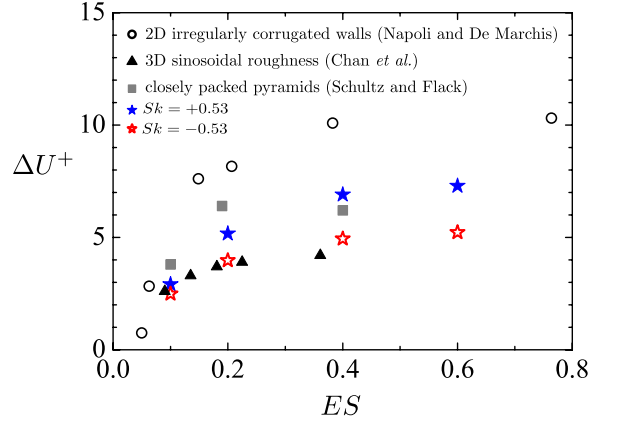


FIG. 5. Roughness function against the  $ES$  value. The LES data for two-dimensional irregularly corrugated walls<sup>5</sup>, the DNS data for three-dimensional sinusoidal roughness<sup>9</sup>, and the experimental data for closely packed pyramids<sup>8</sup> are plotted.

tum equation as follows:

$$\begin{aligned}
 \frac{\partial \langle \bar{u}_i \rangle}{\partial t} + \langle \bar{u}_j \rangle \frac{\partial \langle \bar{u}_i \rangle^f}{\partial x_j} &= -\frac{\varphi}{\rho} \frac{\partial \langle \bar{p} \rangle^f}{\partial x_i} + \frac{\partial}{\partial x_j} \left( \mathbf{v} \frac{\partial \langle \bar{u}_i \rangle}{\partial x_j} \right) \\
 &\quad - \frac{\partial}{\partial x_j} \left( \underbrace{\langle \tilde{u}_i \tilde{u}_j \rangle}_{\mathcal{T}_{ij}} + \underbrace{\langle u'_i u'_j \rangle}_{R_{ij}} \right) \\
 &\quad - \underbrace{\mathbf{v} \frac{\partial \varphi}{\partial x_j}}_{g_i} \frac{\partial \langle \bar{u}_i \rangle^f}{\partial x_j} \\
 &\quad - \underbrace{\frac{1}{\rho S} \int_L \bar{p} n_i d\ell}_{f_{p_i}} - \underbrace{\frac{\mathbf{v}}{S} \int_L \left( -n_k \frac{\partial \tilde{u}_i}{\partial x_k} \right) d\ell}_{f_{v_i}}, \tag{14}
 \end{aligned}$$

where  $L$  represents the obstacle perimeter within an averaging  $x-z$  plane,  $\ell$  represents the circumference length of solid obstacles, and  $n_k$  is its unit normal vector pointing outward from the fluid to solid phase. In addition to the plane-averaged Reynolds stress  $R_{ij}$ , a plane-dispersive covariance  $\mathcal{T}_{ij}$  arises owing to the inhomogeneous nature of the mean flow in the  $x-z$  plane, and is expressed as the product of the mean velocity dispersion  $\tilde{u}_i = \bar{u}_i - \langle \bar{u}_i \rangle^f$ . The other terms representing the roughness effects are the inhomogeneous roughness density term  $g_i$ , the pressure drag term  $f_{p_i}$ , and the viscous drag force term  $f_{v_i}$ . The viscous and pressure drag force terms are expressed as the averaged pressure and viscous stress dispersions, respectively, over the obstacle perimeter at a certain plane. Hence, these terms represent the mean pressure and viscous forces offered by the roughness elements. As  $g_i$  originates from the volume-averaged viscous stress term, it represents the viscous effect arising from the distribution of the plane porosity  $\varphi$ , which exhibits a non-zero contribution only below the maximum roughness crest.

By integrating equation (14) over the wall-normal direction

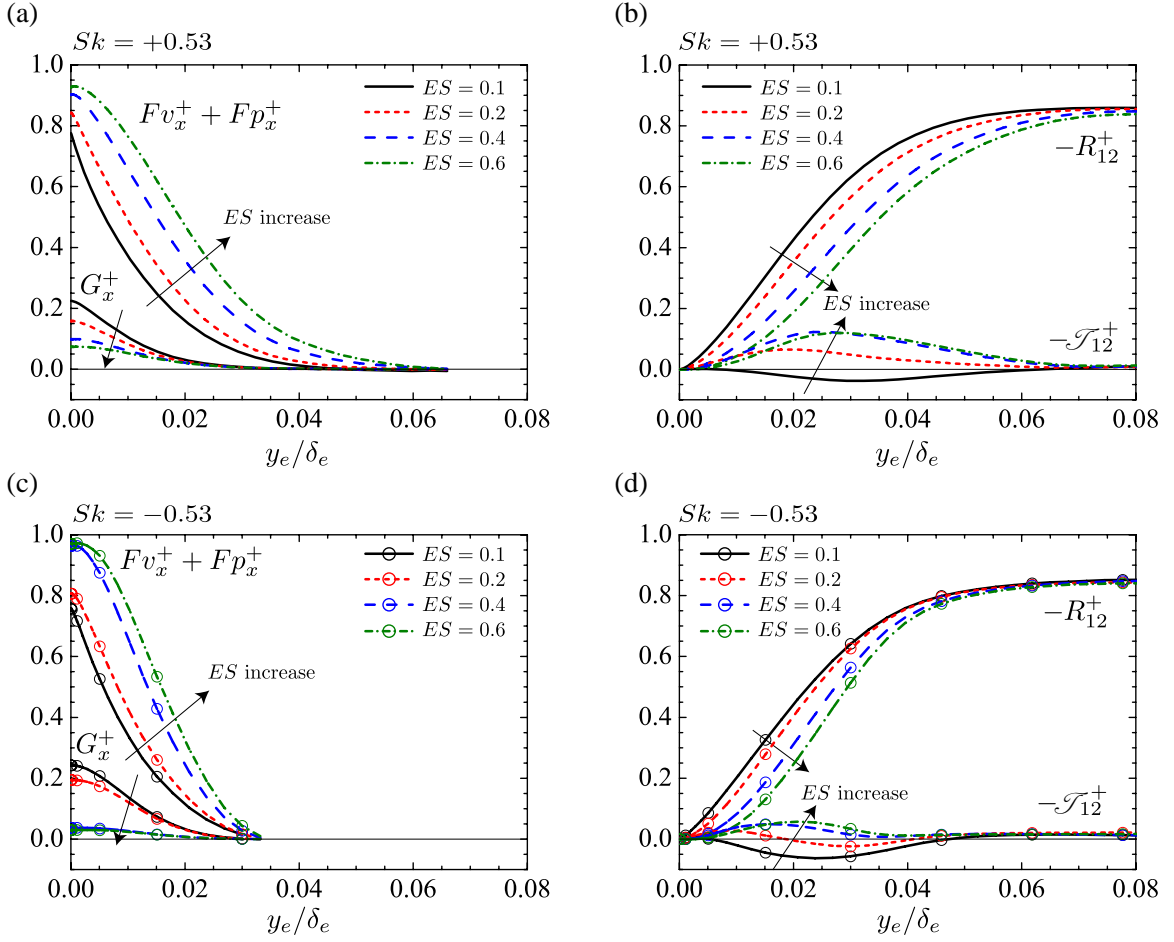


FIG. 6. Momentum budgets: (a) drag force contribution  $Fv_x^+ + Fp_x^+$  and inhomogeneous roughness density contribution  $G_x^+$  for surfaces with  $Sk = +0.53$ , (b) plane-averaged Reynolds stress  $-R_{12}^+$  and plane-dispersive covariance  $-\mathcal{T}_{12}^+$  for surfaces with  $Sk = +0.53$ , (c)  $Fv_x^+ + Fp_x^+$  and  $G_x^+$  for surfaces with  $Sk = -0.53$ , (d)  $-R_{12}^+$  and  $-\mathcal{T}_{12}^+$  for surfaces with  $Sk = -0.53$ .

from 0 to  $y$  and normalizing by  $u_\tau$ , the shear stress balance for the present flow system in non-dimensional form can be derived after some manipulation<sup>18</sup>:

$$1 - \frac{y_e}{\delta_e} = \frac{\partial \langle \bar{u} \rangle^+}{\partial y^+} - R_{12}^+ - \mathcal{T}_{12}^+ + \underbrace{\int_y^{h_p} g_x^+ dy^+}_{G_x^+} + \underbrace{\int_y^{h_p} f_{v_x}^+ dy^+}_{Fv_x^+} + \underbrace{\int_y^{h_p} f_{p_x}^+ dy^+}_{Fp_x^+}, \quad (15)$$

where  $G_x^+$ ,  $Fv_x^+$ , and  $Fp_x^+$  denote the contributory terms of inhomogeneous roughness density, viscous drag, and pressure drag, respectively.

The contributions by the drag force  $Fv_x^+ + Fp_x^+$ , inhomogeneous roughness density  $G_x^+$ , plane-averaged Reynolds shear stress  $-R_{12}^+$ , and plane-dispersive covariance  $-\mathcal{T}_{12}^+$  are shown in Fig. 6. Although the sum of the pressure and viscous drag forces is introduced here, each contribution is further detailed in §VII. In Fig. 6(a) for the peak-dominated surfaces with  $Sk = +0.53$ , the contribution by  $Fv_x^+ + Fp_x^+$  is

substantial below the roughness crest and monotonically increases with the  $ES$  value, whereas  $G_x^+$  decreases with the  $ES$  value. In Fig. 6(b), as the  $ES$  value increases,  $-\mathcal{T}_{12}^+$  increases while  $-R_{12}^+$  decreases. This indicates that  $-\mathcal{T}_{12}^+$  and  $-R_{12}^+$  show a compensating effect, which is consistent with previous observations<sup>18,23</sup>. The reduction in  $-R_{12}^+$  with the  $ES$  value was also found by MacDonald *et al.*<sup>7</sup> who explained that the turbulent near-wall cycle was pushed outward by the wall roughness for densely distributed roughness, which was characterized by the large  $ES$  value. For the valley-dominated surface with  $Sk = -0.53$  in Fig. 6(c) and (d), although the maximum roughness crest measured by  $y_e$  is different from that for the case with  $Sk = +0.53$ , the general trend of the influence of the  $ES$  value on the momentum budget is qualitatively similar to that for the case with  $Sk = +0.53$ .

As can be seen in Fig. 6, the difference in the budget terms between the case with  $Sk = +0.53$  and that with  $Sk = -0.53$  appears to be responsible for the difference in the maximum roughness crest measured by  $y_e$ . To investigate the difference in detail, a comparison with the budget terms between the cases with  $Sk = +0.53$  and  $Sk = -0.53$  is presented in Fig.



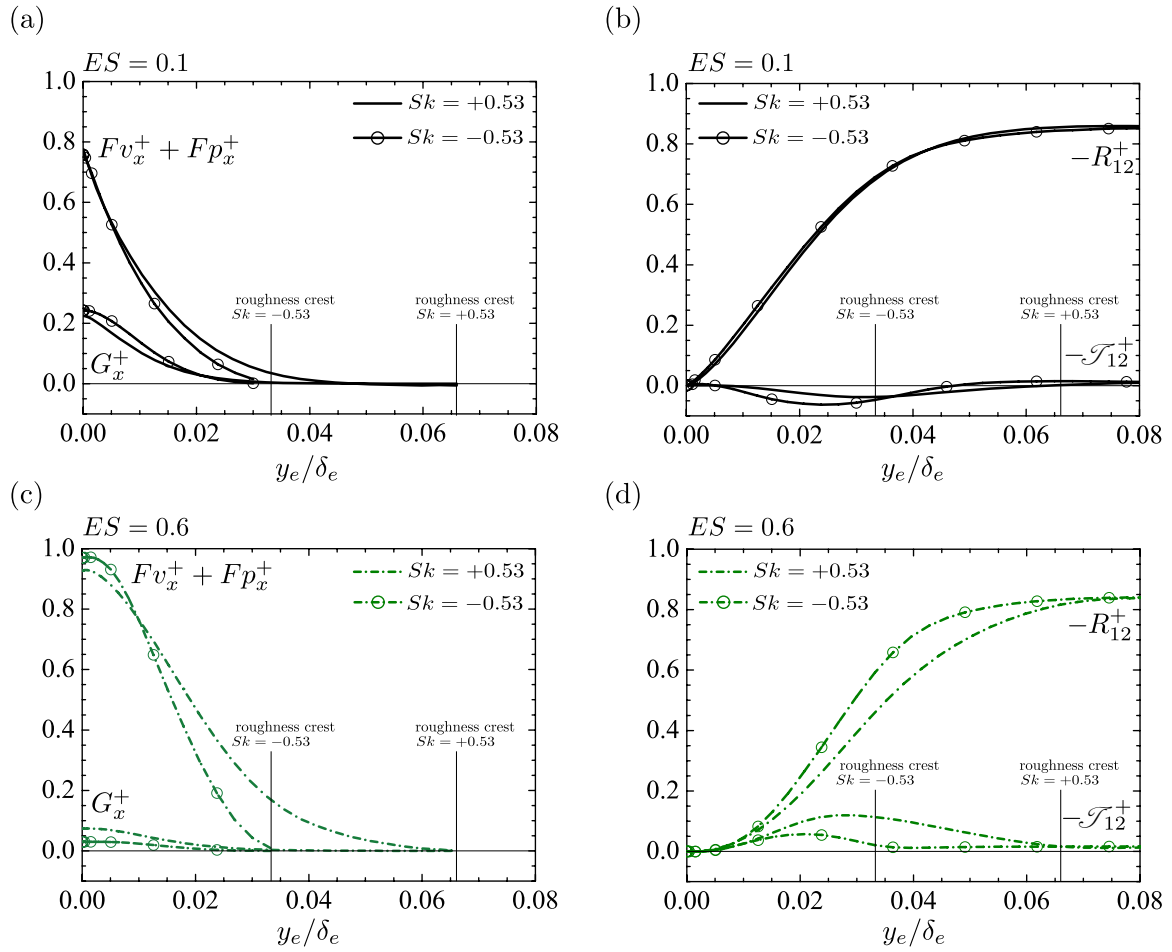


FIG. 7. Comparison of the momentum budgets for cases with  $Sk = +0.53$  and  $Sk = -0.53$ : (a)  $Fv_x^+ + Fp_x^+$  and  $G_x^+$  for surfaces with  $ES = 0.1$ , (b)  $-R_{12}^+$  and  $-\mathcal{T}_{12}^+$  for surfaces with  $ES = 0.1$ , (c)  $Fv_x^+ + Fp_x^+$  and  $G_x^+$  for surfaces with  $ES = 0.6$ , (d)  $-R_{12}^+$  and  $-\mathcal{T}_{12}^+$  for surfaces with  $ES = 0.6$ .

7. It is evident from Fig. 7(a) and (c) that the drag force,  $Fv_x^+ + Fp_x^+$ , below the roughness crest ( $0.03 < y_e < 0.07$ ) is significantly larger for the case with  $ES = 0.6$  than that for the case with  $ES = 0.1$ , which enlarges the difference in the momentum budgets between the case with  $Sk = +0.53$  and that with  $Sk = -0.53$  ( $ES = 0.6$ ) as shown in Fig. 7(c). As in  $Fv_x^+ + Fp_x^+$ , it is found from Fig. 7(b) and (d) that the difference in the second moments ( $-R_{12}^+$  and  $-\mathcal{T}_{12}^+$ ) is more pronounced for cases with  $ES = 0.6$  below the roughness crest ( $0.03 < y_e < 0.07$ ).

The physical interpretation of the influence of the  $ES$  values on the second moments can be provided by the spatial distribution of  $-\tilde{u}\tilde{v}^+$  and  $-\overline{u'v'^+}$  in the  $x-z$  plane for the case with  $Sk = +0.53$ . Figures 8 and 10 depict the contour maps of  $-\tilde{u}\tilde{v}^+$  and  $-\overline{u'v'^+}$ , respectively, at  $y_e/\delta_e = 0.02$ , where the effects of the  $ES$  values on  $R_{12}^+$  and  $\mathcal{T}_{12}^+$  are pronounced, as shown in Fig. 6(b). Also shown are the contour lines to indicate the high-speed region and the backward flow region. The contour maps for  $ES = 0.6$  in Figs. 10(b) and 8(b) are enlarged by a factor of 6 for a comparison with those for  $ES = 0.1$ . The dispersive shear stress of  $-\tilde{u}\tilde{v}^+$  for the case with  $ES = 0.1$  in

Fig. 8 (a) is significant but the averaged value over the  $x-z$  plane is considerably small ( $-\langle\tilde{u}\tilde{v}^+\rangle = -0.025$ ), thereby suggesting that there is almost no correlation between  $\tilde{u}$  and  $\tilde{v}$  for the case with  $ES = 0.1$ . In contrast, for the case with  $ES = 0.6$  in Fig. 8 (b), although the magnitude of  $-\tilde{u}\tilde{v}^+$  is not significant,  $-\tilde{u}\tilde{v}^+$  tends to be positive for the high-speed and backward flow regions. This means that there is a strong negative correlation between  $\tilde{u}$  and  $\tilde{v}$  in the regions where the downwash flow of the high-speed fluid and upwash flow of the backward flow region frequently occur. The dominance of the negative dispersive shear stress for the case with  $ES = 0.6$  can be statistically proven by Fig. 9 where the joint probability density function of  $\tilde{u}^+$  and  $\tilde{v}^+$  is depicted. For the case with  $ES = 0.6$  in Fig. 9 (b), the occurrence of strong positive and negative  $\tilde{u}^+$  is found to be less frequent than that for the case with  $ES = 0.1$  in Fig. 9 (a). However, it is apparent in Fig. 9(b) that, for the case with  $ES = 0.6$ , there is a distinct negative correlation between  $\tilde{v}^+$  and  $\tilde{u}^+$ , which produces the negative dispersion shear stress. This observation clearly explains the reason why  $-\mathcal{T}_{12}^+$  increases with the  $ES$  values in Fig. 6(b).

The Reynolds shear stress  $-\overline{u'v'^+}$  in Fig. 10 is generally pos-

itive, as in smooth-wall turbulence, but significantly reduced near the roughness elements. Interestingly,  $-\overline{u'v'}$  is considerably larger for the case with  $ES = 0.1$  (Fig. 10(a)) than for the case with  $ES = 0.6$  (Fig. 10(b)) despite the fact that the skewness factor and roughness height scales are consistent. This difference can be reasonably explained from a snapshot of the instantaneous wall-normal vortex fluctuations shown in Fig. 11. In Fig. 11(a), quasi-streamwise vortices, which are similar to those in smooth-wall turbulence<sup>38</sup>, develop for the case with  $ES = 0.1$  because of the sufficient scale separation between the roughness wavelength and the turbulent length scale. However, for the case with  $ES = 0.6$  in Fig. 11(b), as the roughness wavelength in the  $x$ - and  $z$ - directions is reduced by a factor of 6, the turbulent length scale is observed to be comparable to the roughness length scale. Consequently, the roughness elements effectively prevent the development of the quasi-streamwise vortices. A similar observation of the effects of the roughness on quasi-streamwise vortices was made by Kuwata and Kawaguchi<sup>32</sup>. Their budget term analysis suggested that the breakdown of quasi-streamwise vortices was a representation of additional energy dissipation due to a role of the drag force offered by the roughness elements. In other words, the turbulence generation tended to be attenuated with an increase in the drag force. This reasonably explains why  $-\overline{u'v'}$  is smaller for the case with  $ES = 0.6$ , in which the drag force contribution  $Fv_x^+ + Fp_x^+$  is substantial within the rough wall, as shown in Fig. 6(a).

## VII. CONTRIBUTION OF THE ROUGHNESS FUNCTION

Although the previous section concentrates on the influence of  $ES$  and  $Sk$  values on momentum transfer, it is still not clear how these momentum budget terms contribute to an increase in  $\Delta U^+$ . Hence, in this section we quantify the effects on  $\Delta U^+$  starting from Eq. (15). The double-average momentum equation can be rewritten as follows:

$$\frac{\partial \varphi \langle \bar{u} \rangle^{f+}}{\partial y^+} = 1 - \frac{y_e}{\delta_e} + R_{12}^+ + \mathcal{T}_{12}^+ - G_x^+ - Fv_x^+ - Fp_x^+. \quad (16)$$

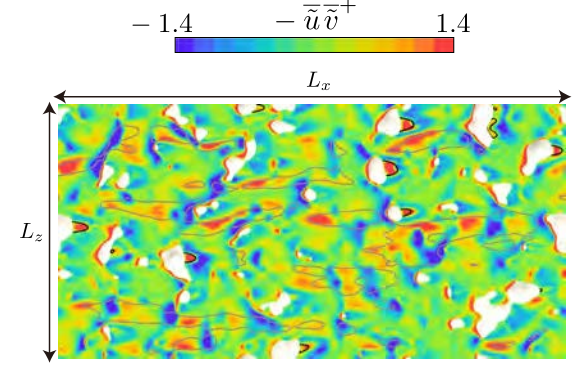
The left-hand side of Eq. (16) can be rewritten by using the definition of  $y_e$  in Eq. (12) as follows:

$$\frac{\partial \varphi \langle \bar{u} \rangle^{f+}}{\partial y^+} = \frac{\partial \varphi^2 \langle \bar{u} \rangle^{f+}}{\partial y_e^+} - \langle \bar{u} \rangle^{f+} \frac{\partial \varphi}{\partial y^+}. \quad (17)$$

By integrating Eq. (16) with Eq. (17) over the wall-normal direction from 0 to  $y_e^+$ , the mean velocity can be written in terms of the momentum budget terms as follows:

$$\begin{aligned} \varphi^2 \langle \bar{u} \rangle^{f+}(y_e^+) &= y_e^+ \left( 1 - \frac{1}{2} \frac{y_e^+}{Re_\tau} \right) + \int_0^{y_e^+} (R_{12}^+ + \mathcal{T}_{12}^+) dy_e^+ \\ &\quad - \int_0^{y_e^+} \left( G_x^+ - \langle \bar{u} \rangle^{f+} \frac{\partial \varphi}{\partial y^+} \right) dy_e^+ \\ &\quad - \int_0^{y_e^+} Fv_x^+ dy_e^+ - \int_0^{y_e^+} Fp_x^+ dy_e^+. \end{aligned} \quad (18)$$

(a)  $ES = 0.1$



(b)  $ES = 0.6$

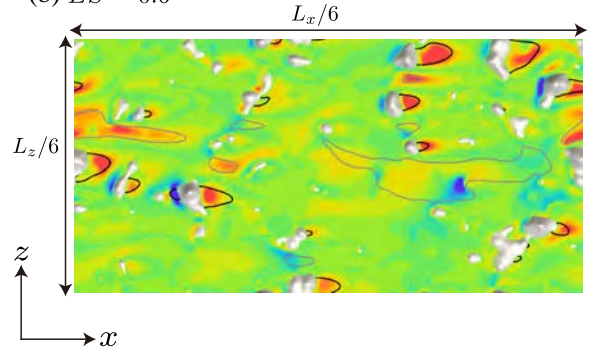


FIG. 8. Contour map of the dispersive shear stress  $-\tilde{u}\tilde{v}^+$  in the  $x-z$  plane at  $y_e/\delta_e = 0.02$ : (a) case with  $ES = 0.1$  and  $Sk = +0.53$  and (b) case with  $ES = 0.6$  and  $Sk = +0.53$ . The regions enclosed by gray lines indicate the high-speed flow regions of  $\bar{u}^+ > 0.8\bar{u}_{max}^+$ , where  $\bar{u}_{max}^+$  is the maximum velocity in the plane; those enclosed by black lines are the backward flow regions of  $\bar{u}^+ < 0$ .

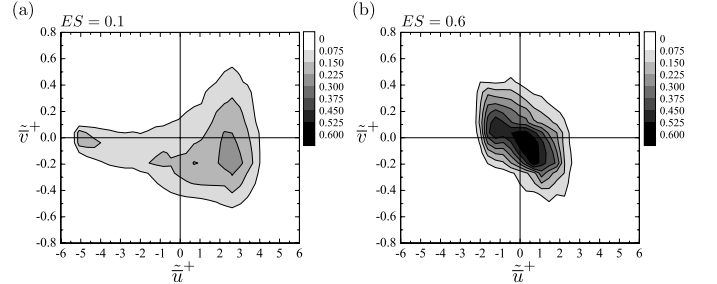


FIG. 9. Joint probability density function of the streamwise and wall-normal velocity dispersions: (a) case with  $ES = 0.1$  and  $Sk = +0.53$  and (b) case with  $ES = 0.6$  and  $Sk = +0.53$ .

The corresponding equation for the smooth-wall case can be derived in a similar fashion as follows:

$$\langle \bar{u} \rangle^{f+}(y_e^+) = y_e^+ \left( 1 - \frac{1}{2} \frac{y_e^+}{Re_\tau} \right) + \int_0^{y_e^+} R_{12}^+ dy_e^+. \quad (19)$$

Note that  $y_e = y$  for the smooth-wall case because  $\varphi = 1$ . Subtracting the mean velocity for the rough-wall case in Eq. (18) from that for the smooth-wall case in Eq. (19) at  $y_e^+ = 100$ , we can derive an expression for  $\Delta U^+$  in terms of several con-

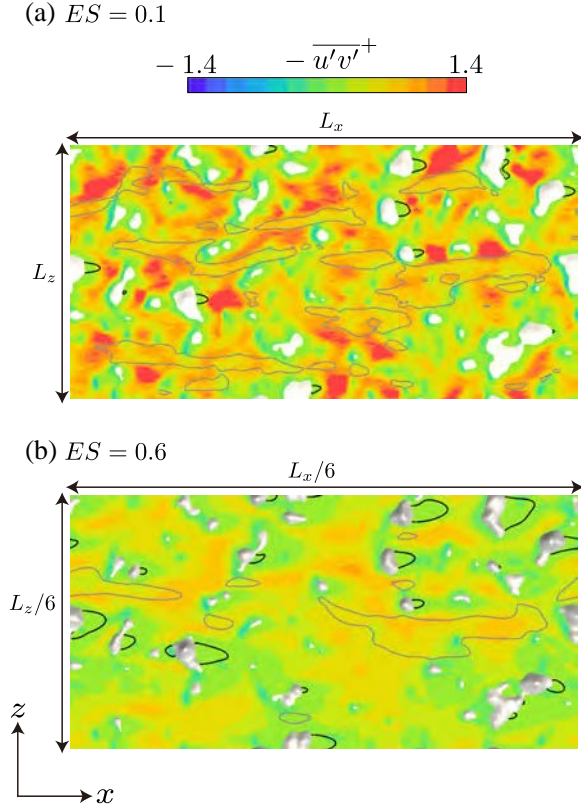


FIG. 10. Contour map of the Reynolds shear stress,  $-\overline{u'v'}^+$ , in the  $x-z$  plane at  $y_e/\delta_e = 0.02$ : (a) case with  $ES = 0.1$  and  $Sk = +0.53$  and (b) case with  $ES = 0.6$  and  $Sk = +0.53$ . The regions enclosed by gray lines indicate the high-speed flow regions of  $\bar{u}^+ > 0.8\bar{u}_{max}^+$ , where  $\bar{u}_{max}^+$  is the maximum velocity in the plane; those enclosed by black lines are the backward flow regions of  $\bar{u}^+ < 0$ .

tributors, as follows:

$$\begin{aligned} \Delta U^+ &= \Delta U_{sm}^+ + \Delta U_{dv}^+ + \Delta U_{dp}^+ + \Delta U_{ir}^+, \\ \Delta U_{sm}^+ &= \int_0^{100} R_{12}^+ dy_e^+ \Big|_{smooth} - \int_0^{100} (R_{12}^+ + \mathcal{T}_{12}^+) dy_e^+ \Big|_{rough}, \\ \Delta U_{dv}^+ &= \int_0^{h_{pe}^+} F v_x^+ dy_e^+ \Big|_{rough}, \\ \Delta U_{dp}^+ &= \int_0^{h_{pe}^+} F p_x^+ dy_e^+ \Big|_{rough}, \\ \Delta U_{ir}^+ &= \int_0^{h_{pe}^+} \left( G_x^+ - \langle u \rangle^{f+} \frac{\partial \phi}{\partial y^+} \right) dy_e^+ \Big|_{rough}, \end{aligned} \quad (20)$$

where  $h_{pe}$  is the maximum roughness crest evaluated by  $y_e$ , and the contributors  $\Delta U_{sm}^+$ ,  $\Delta U_{dv}^+$ ,  $\Delta U_{dp}^+$ , and  $\Delta U_{ir}^+$  represent the effects of the second moment, viscous drag, pressure drag, and inhomogeneous roughness terms, respectively. It is to be noted that MacDonald *et al.*<sup>7</sup> as well as Jelly and Busse<sup>23</sup> derived similar expressions for  $\Delta U$ ; however, they did not separately consider the contributions of  $\Delta U_{dv}^+$ ,  $\Delta U_{dp}^+$ , and  $\Delta U_{ir}^+$ , which have a dominant effect below the roughness crest. The contributors in Eq.(20) are shown in Fig. 12. The figure confirms that  $\Delta U^+$  is dominated by the pressure and viscous drag

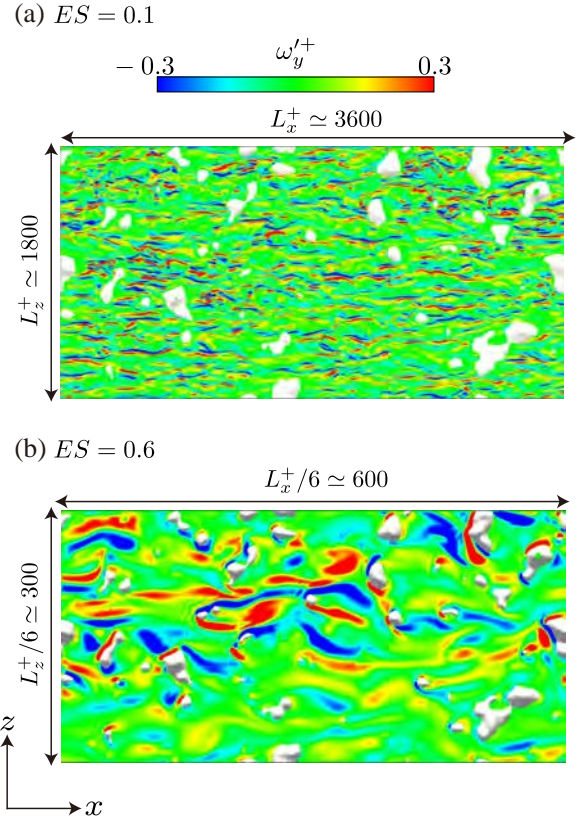


FIG. 11. Snapshot of the wall-normal vorticity fluctuations in the  $x-z$  plane at  $y_e/\delta_e = 0.02$ : (a) case with  $ES = 0.1$  and  $Sk = +0.53$  and (b) case with  $ES = 0.6$  and  $Sk = +0.53$ .

effects ( $\Delta U_{dv}^+$  and  $\Delta U_{dp}^+$ ), whereas the second moment  $\Delta U_{sm}^+$  exhibits a negative contribution. The viscous drag  $\Delta U_{dv}^+$  dominates  $\Delta U^+$  when  $ES = 0.1$  but does not increase with the  $ES$  value; however,  $\Delta U_{dp}^+$  dramatically increases with the  $ES$  value, which supports the findings by Napoli *et al.*<sup>5</sup> as well as Leonardi and Castro<sup>13</sup>, who showed that the contribution of the pressure drag to the friction drag increased with an increasing  $ES$  value. A comparison between the results for the case with  $Sk = +0.53$  and that with  $Sk = -0.53$  suggests that  $\Delta U_{dp}^+$  is consistently larger for the peak-dominated surface with  $Sk = +0.53$  and that this is the dominant mechanism leading to the larger  $\Delta U^+$  for the peak-dominated surface. The negative contribution of  $\Delta U_{sm}^+$  is more significant as the  $ES$  value increases, i.e., the friction drag associated with  $-(R_{12}^+ + \mathcal{T}_{12}^+)$  is less significant than that for the smooth-wall case and decreases with the  $ES$  value. This is principally due to the significant reduction in  $-R_{12}^+$  with the  $ES$  value, as observed in Fig. 6(b) and (d). Another notable observation from Fig. 12 is that the negative contribution by  $\Delta U_{sm}^+$  for cases with  $ES \geq 0.4$  partly cancels the positive contributions by  $\Delta U_{dv}^+$  and  $\Delta U_{dp}^+$ , which explains why  $\Delta U^+$  does not significantly increase with the  $ES$  value for the roughness regime ( $ES \geq 0.4$ ) as observed in Fig. 5.



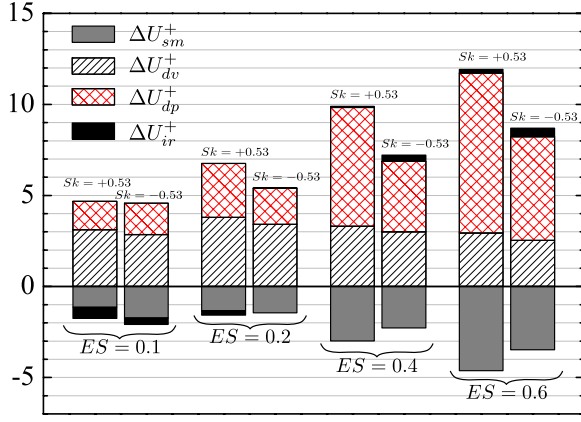


FIG. 12. Contribution of the roughness function. The contributory terms:  $\Delta U_{sm}^+$  is the second moment,  $\Delta U_{dv}^+$  is the viscous drag,  $\Delta U_{dp}^+$  is the pressure drag, and  $\Delta U_{ir}^+$  represents inhomogeneous roughness, which are expressed as Eq. (20).

### VIII. DRAG FORCE

In this section, we describe an attempt to obtain a physical understanding of the drag force effects expressed as  $\Delta U_{dv}^+$  and  $\Delta U_{dp}^+$ . The viscous drag contributor, which is expressed as the double integral of  $f_{vx}^+$ , can be transformed by partial integration as follows:

$$\begin{aligned} \Delta U_{dv}^+ &= \int_0^{h_p^+} \left( \int_{y^+}^{h_p^+} f_{vx}^+ dy^+ \right) dy_e^+ \\ &= \int_0^{h_p^+} \frac{dy_e^+}{dy^+} \left( \int_{y^+}^{h_p^+} f_{vx}^+ dy^+ \right) dy^+ \\ &= \left[ y_e^+ \left( \int_{y^+}^{h_p^+} f_{vx}^+ dy^+ \right) \right]_0^{h_p^+} \\ &\quad - \int_0^{h_p^+} y_e^+ \frac{d}{dy^+} \left( \int_{y^+}^{h_p^+} f_{vx}^+ dy^+ \right) dy^+, \quad (21) \end{aligned}$$

As  $y_e^+ = 0$  when  $y^+ = 0$ , the definite integral in the first term on the right-hand side of Eq. (21) goes to zero as follows:

$$\left[ y_e^+ \left( \int_{y^+}^{h_p^+} f_{vx}^+ dy^+ \right) \right]_0^{h_p^+} = 0. \quad (22)$$

In addition, the differentiation of the definite integral in the second term on the right-hand side of Eq. (21) can be written as

$$\frac{d}{dy^+} \left( \int_{y^+}^{h_p^+} f_{vx}^+ dy^+ \right) = -f_{vx}^+. \quad (23)$$

Substituting Eqs.(22) and (23) into Eq. (21), the contribution by the viscous drag force can be simply expressed as the weighted integral of  $f_{vx}^+$ , as follows:

$$\Delta U_{dv}^+ = \int_0^{h_p^+} y_e^+ f_{vx}^+ dy^+. \quad (24)$$

The contribution by the pressure drag can be expressed in a similar fashion as follows:

$$\Delta U_{dp}^+ = \int_0^{h_p^+} y_e^+ f_{px}^+ dy^+. \quad (25)$$

The interesting implication from Eqs.(24) and (25) is that  $\Delta U^+$  is not solely affected by the magnitude of  $f_{px}^+$  or  $f_{vx}^+$  itself; rather, the product of the effective distance  $y_e$  and the drag force is the key factor. First, the dominant mechanism of the increase in  $\Delta U^+$ , namely the pressure drag effects, is shown against  $y/h$  and  $y^* = (y - h_m)/h_{rms}$  in Fig. 13. In Fig. 13 (a) and (b), as the  $ES$  value increases,  $f_{px}^+$  progressively increases above the mean location of the surface ( $y^* > 0$ ) but decreases near the bottom ( $-2 < y^* < -1$ ). The primary reason for the increase in  $f_{px}^+$  is an increase in the wetted area of the rough surface with the  $ES$  value: the wetted area is proportional to the  $ES$  value for the surfaces considered in this study, and thus it increases by a factor of 6 as  $ES$  increases from 0.1 to 0.6. It is worth noting that the effective distance  $y_e$  increases with  $y/\delta$  as shown in Fig. 13(a) and (b). Hence,  $f_{px}^+$  near the bottom wall plays a less important role in  $y_e^+ f_{px}^+$  but is more influential as it moves toward the roughness crest. This can be clearly confirmed from Fig. 13 (c) and (d): the decrease in  $f_{px}^+$  with the  $ES$  value below the mean location of the surface ( $y^* < 0$ ) is less visible for  $y_e^+ f_{px}^+$ , while the increase in  $f_{px}^+$  with the  $ES$  value near the roughness crest is enlarged. Consequently, it can be concluded that  $\Delta U^+$  increases with the  $ES$  value, owing to the increase in  $f_{px}^+$  above the mean location of the surface.

A comparison of  $f_{px}^+$  for the case with  $Sk = +0.53$  (Fig. 13(a)) against the case with  $Sk = -0.53$  (Fig. 13(b)) confirms that the location where  $f_{px}^+$  reaches the maximum peak value is almost the same when scaled by  $y^*$  as  $0 < y^* < 1$ , while the maximum peak value of  $f_{px}^+$  is found to be larger for the valley-dominated surface ( $Sk = -0.53$ ) when the  $ES$  value is the same. However,  $y_e^+$  against  $y/\delta$  is considerably larger for cases with  $Sk = +0.53$ ; thus, the difference in the maximum peak value of  $f_{px}^+$  becomes considerably smaller for  $y_e^+ f_{px}^+$ , as shown in Fig. 13(c) and (d). As a result, the integral of  $y_e^+ f_{px}^+$  expressed as Eq. (25) becomes larger for the surface with  $Sk = +0.53$ . This is the reason why the contribution of the pressure drag  $\Delta U^+$  is larger for the surface with  $Sk = +0.53$ .

The viscous drag  $f_{vx}^+$  and the weighted viscous drag  $y_e^+ f_{vx}^+$  are also presented in Fig. 14. Figure 14 (a) and (b) confirms that, in contrast with the pressure drag, the maximum peak of  $f_{vx}^+$  generally decreases with the  $ES$  value despite the increase in the wetted area of the rough surface. In the region of  $-2 < y^* < 1$ ,  $f_{vx}^+$  rapidly decreases with the  $ES$  value and eventually exhibits a negative value for the cases with  $ES = 0.4$  and  $0.6$ . This trend is consistent irrespective of the  $Sk$  value. However, as the effective distance  $y_e^+$  is not significant in those regions, the significant reduction of  $f_{vx}^+$  with the  $ES$  value is less influential for  $y_e^+ f_{vx}^+$ , which explains why the contribution of the roughness  $\Delta U_{dp}^+$  is not significantly affected by the  $ES$  value, as shown in Fig. 12.

Finally, we focus on the influence of  $ES$  values on the mean flow structure, which is closely related to the drag force terms

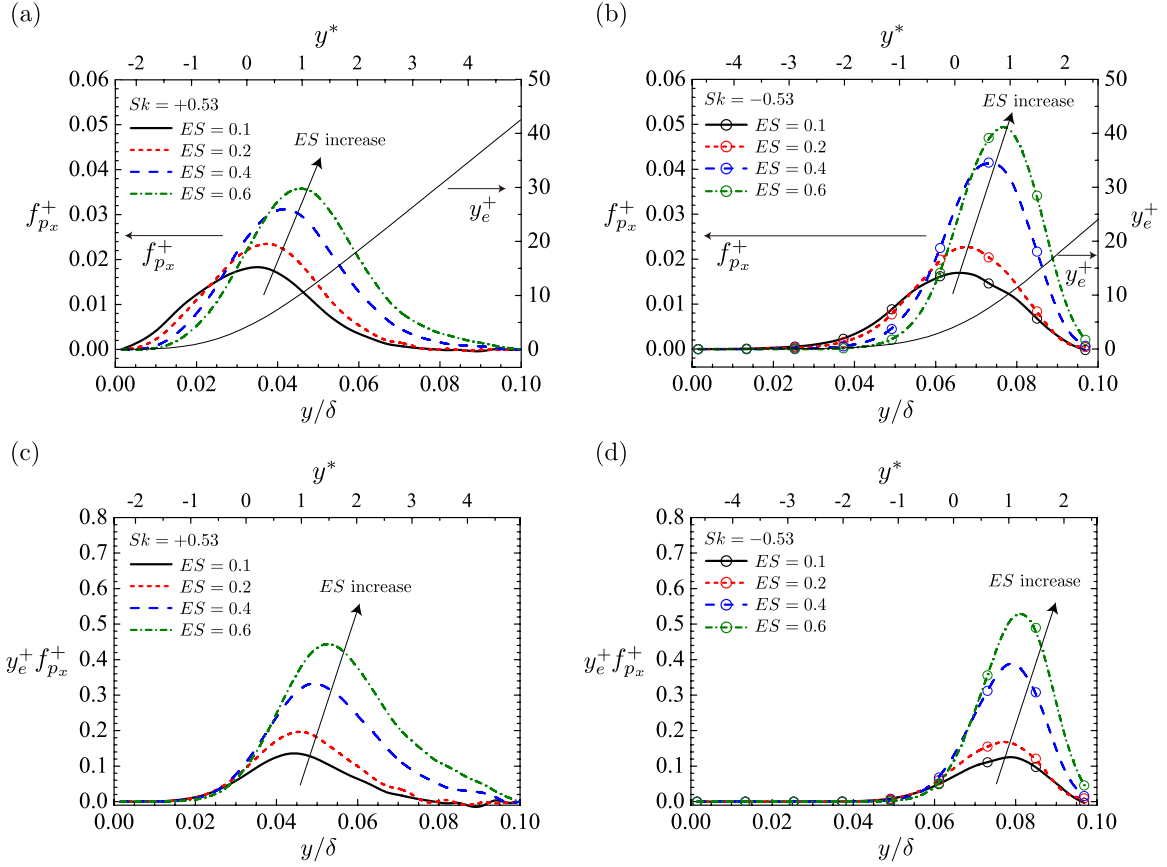


FIG. 13. Profiles of the pressure drag against  $y/h$  and  $y^* = (y - h_m)/h_{rms}$ : (a) pressure drag  $f_{p_x}^+$  for cases with  $Sk = +0.53$ , (b)  $f_{p_x}^+$  for cases with  $Sk = -0.53$ , (c) weighted pressure drag  $y_e^+ f_{p_x}^+$  for cases with  $Sk = +0.53$ , and (d)  $y_e^+ f_{p_x}^+$  for cases with  $Sk = -0.53$ . Thin solid lines in (a) and (b) indicate the profiles of  $y_e^+$ .

$f_{v_x}^+$  and  $f_{p_x}^+$ . The PDF of  $\bar{u}^+$  below the roughness crest is shown for the case with  $Sk = +0.53$  in Fig. 15. As aforementioned, the influence of the  $ES$  value on the drag force terms is similar irrespective of the  $Sk$  values, and only the results for the case with  $Sk = +0.53$  are shown here (we have confirmed that the same conclusion can be drawn for the case with  $Sk = -0.53$ ). In Fig. 15(a) and (b), the most notable difference between the results for the cases with  $ES = 0.1$  and  $ES = 0.6$  is the probability density of the backward flow near the location of the mean surface  $y^* = -0.12$ : although a weak backward flow with  $-1 < \bar{u}^+ < 0$  occurs in both cases, the PDF of  $-1 < \bar{u}^+ < 0$  significantly increases for the case with  $ES = 0.6$ , indicating an increase in the weak backward flow region. For the case with  $ES = 0.6$ , the PDFs at  $y^* = -0.12$  and  $0.53$  exhibit peaks around  $\bar{u}^+ \simeq 0$ , indicating that the flow field is dominated by the dead water region where the mean positive or negative flow is sufficiently weak to interact with outer flows. In the dead water region, the wall-shear stress is very minimal and sometimes shows a negative value; thus,  $f_{v_x}^+$  for the case with  $ES = 0.6$  exhibits a smaller value than that for the case with  $ES = 0.1$  near the location of the mean surface, as observed in Fig. 14(a). Further, in this region the mean pressure dispersion is considerably small, which may explain the reason why  $f_{p_x}^+$  below the location of the mean

surface decreases with the  $ES$  value in Fig. 13(a).

The increase in the dead flow regions near the location of the mean surface can be confirmed by a contour map of the streamwise mean velocity at  $y^* = -0.12$  in Fig. 16 together with contour lines of  $\bar{u} = 0$ . It is immediately observed that the weak backward flow regions ( $-2 < \bar{u}^+ < 0$ ) associated with the recirculating bubbles behind the roughness elements are merged and significantly extended for the case with  $ES = 0.6$  in Fig. 16(b), while the backward flow regions for the case with  $ES = 0.1$  in Fig. 16(a) are limited to the region immediately behind the roughness elements. This is principally due to insufficient scale separation between the roughness wavelength and the turbulent length scale for the dense roughness ( $ES = 0.6$ ), as shown in Fig. 11(b). In this situation, the turbulent vortices do not penetrate the dead water region, and the stable weak vortices within the roughness, which are driven by the skimming flow above the roughness crest, are not disrupted by the turbulent vortices. Therefore, the mean flow structure is analogous to that seen in  $d$ -type roughness<sup>14,39,40</sup>.

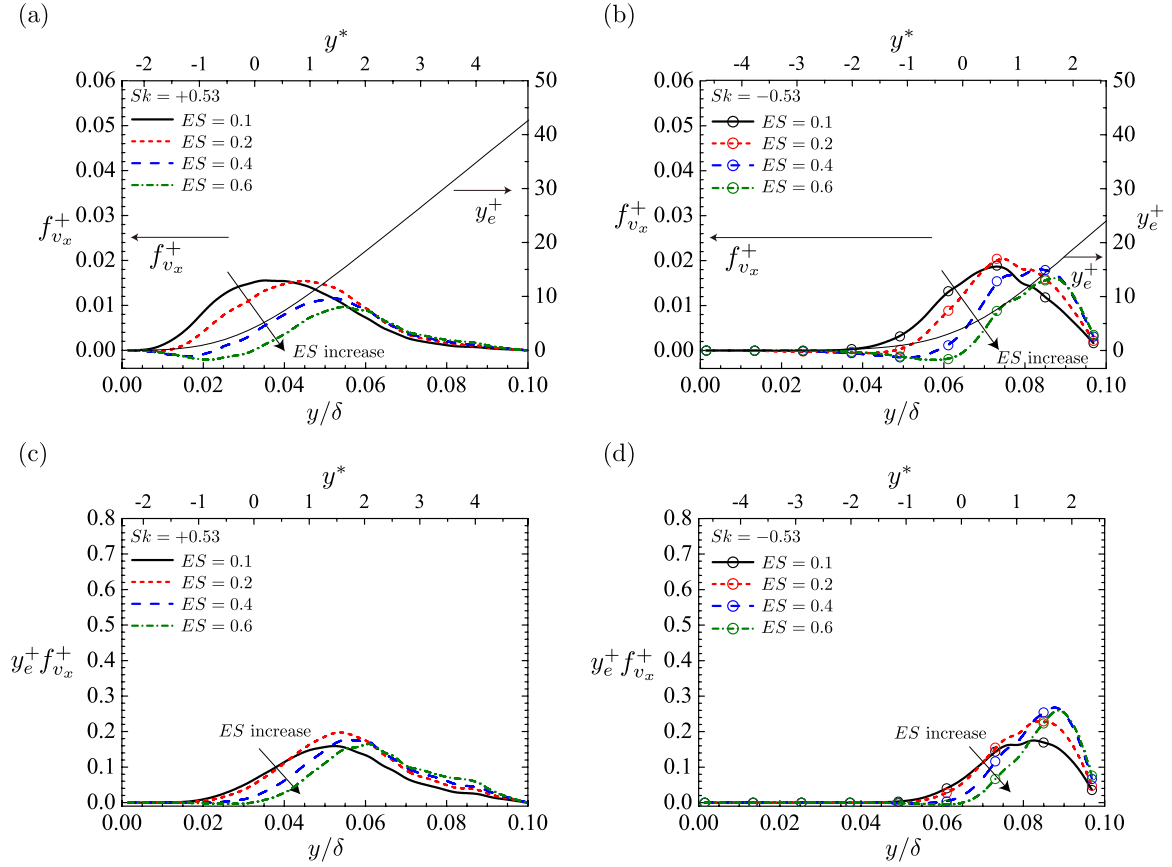


FIG. 14. Profiles of the viscous drag against  $y/h$  and  $y^* = (y - h_m)/h_{rms}$ : (a) viscous drag  $f_{v_x}^+$  for cases with  $Sk = +0.53$ , (b)  $f_{v_x}^+$  for cases with  $Sk = -0.53$ , (c) weighted viscous drag  $y_e^+ f_{v_x}^+$  for cases with  $Sk = +0.53$ , and (d)  $y_e^+ f_{v_x}^+$  for cases with  $Sk = -0.53$ . Thin solid lines in (a) and (b) indicate the profiles of  $y_e^+$ .

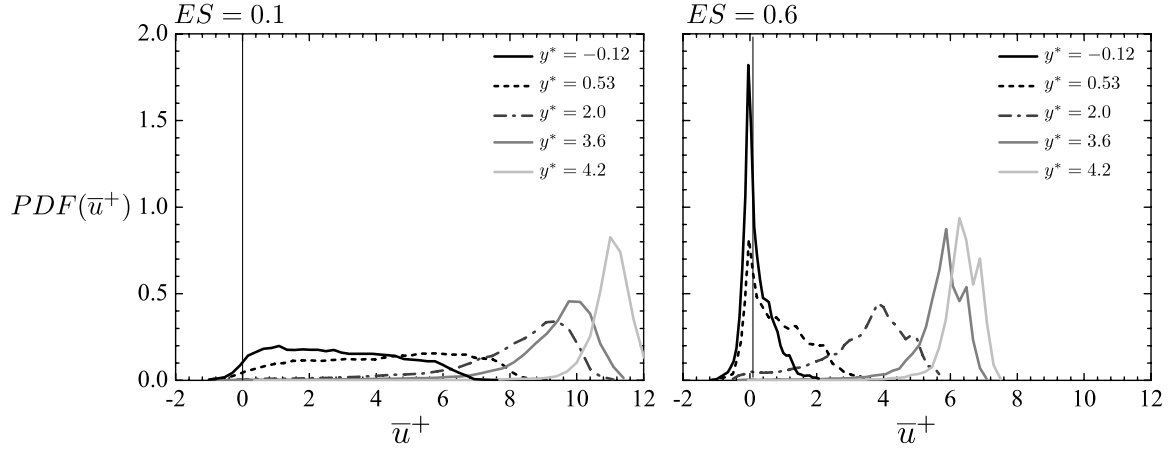


FIG. 15. PDF of the streamwise mean velocity,  $\bar{u}^+$ , in the  $x-z$  plane at different  $y^*$  locations for cases with  $Sk = +0.53$ .

## IX. CONCLUSIONS

The influence of two important geometrical parameters for rough surfaces, namely, the effective slope  $ES$  and skewness factor  $Sk$ , on rough wall turbulence was analyzed by DNSs of turbulence over systematically varied three-dimensional ir-

regular rough surfaces. We numerically generated the rough surfaces in which the  $ES$  and  $Sk$  values were systematically varied in the range of  $Sk = \pm 0.53$  and  $0.1 \leq ES \leq 0.6$ , while roughness height scales remained fixed.

We confirmed that the roughness function  $\Delta U^+$  is larger for a peak-dominated surface with  $Sk = +0.53$  than that for a



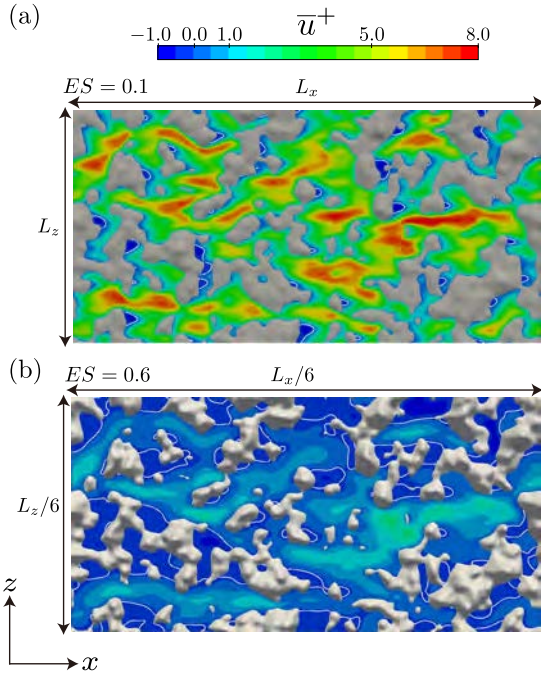


FIG. 16. Contour map of the streamwise mean velocity  $\bar{u}^+$  in the  $x-z$  plane at  $y_e/\delta_e = 0.02$  for cases with  $Sk = +0.53$ : (a) for case with  $ES = 0.1$  and (b) for case with  $ES = 0.6$ . The regions enclosed by white lines are the backward flow regions of  $\bar{u}^+ < 0$ .

valley-dominated surface with  $Sk = -0.53$ . The dependence of  $Sk$  on  $\Delta U^+$  increases as the  $ES$  value increases. As for the influence of  $ES$ , it was found that  $\Delta U^+$  steeply increases when  $0.1 \leq ES \leq 0.4$  whereas the increase slows down when  $ES \geq 0.4$ .

The physical mechanism behind the increase in  $\Delta U^+$  was discussed by analyzing the spatial- and Reynolds-averaged momentum equation. For a surface with  $ES \leq 0.2$ , the viscous drag dominantly contributes to  $\Delta U^+$ , whereas the contribution by the pressure drag progressively increases with the  $ES$  value and becomes dominant when  $ES \geq 0.4$ . Meanwhile, when  $ES \geq 0.4$ , as there is no sufficient scale separation between the roughness wavelength and the turbulent length scale, the wall roughness prevents the formation of quasi-streamwise elongated vortices that suppress the turbulent near-wall cycles, thereby decreasing the Reynolds shear stress. This acts as a negative contribution to  $\Delta U^+$ , and the two aforementioned competing effects weaken the dependence of the  $ES$  value on  $\Delta U^+$  when  $ES \geq 0.4$ . As for the influence of the  $Sk$  value, the pressure drag is consistently larger for the surface with  $Sk = +0.53$ ; this is the dominant mechanism leading to the larger  $\Delta U^+$  for the surface with  $Sk = +0.53$ .

The drag force effects, which have a dominant impact on the increase in  $\Delta U^+$ , were further analyzed. The contribution of  $\Delta U^+$  by the drag force was found to be the integral of the weighted drag force, which is expressed as the product of the drag force and effective distance from within the rough walls. Because the effective distance increases as it approaches the roughness crest, the drag force near the roughness crest provides a larger contribution to the increase in  $\Delta U^+$ . The ef-

fective distance, which depends on the geometry of the rough surfaces, is generally larger for a surface with  $Sk = +0.53$ , thus leading to the increase in the drag force contribution to  $\Delta U^+$  for the surface with  $Sk = +0.53$ . This is the main reason why peak-dominated surfaces with a positive  $Sk$  yield larger  $\Delta U^+$ .

The present strategy for varying surface characteristics was found to be meaningful in investigating the isolated effects of the effective slope  $ES$  and skewness factor  $Sk$ . However, the present study only considers a single relatively low friction Reynolds number of 600, owing to limitations in computational resources; thus, the simulated flows are generally in the transitionally rough regime. Further analysis in the fully rough regime will be required for full appreciation of the effects. This will be accomplished through experiments or minimal flow simulations as in Macdonald *et al.*<sup>41</sup>.

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## DATA AVAILABILITY

The data that support the findings of this study are available from the corresponding author upon reasonable request.

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